


GLOBAL POSITIONING SYSTEM (GPS) PERFORMANCE

QUARTERLY REPORT 3 (JULY TO SEPTEMBER
2023)


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1. INTRODUCTION

1.1. PURPOSE

This document presents the results of the GPS SPS performance assessment for the period of July to September 2023. The objectives of the study are to compare the measured performance against US DoD SPS performance specification [RD.1], covering the following parameters [AD.1]:

- SPS SiS Accuracy,
- SPS SiS Integrity,
- SPS SiS Continuity,
- SPS SiS Availability,
- PDOP Availability,
- SPS Position Service Availability and
- SPS Position Service Accuracy.

It also includes NANU analysis. The performance is analysed using raw data recorded at the OSi site SLGO.

1.2. DOCUMENT OVERVIEW

This document is arranged in the following sections:

- **Section 1**, the current section, describes the purpose, scope and structure of the document and lists the reference documents.
- **Section 2** gives an introduction to the activity, including performance specification and assessment methodology and assumptions;
- **Section 3** contains an assessment of performance against GPS SPS performance standards;
- **Section 4** provides an analysis of the NANUs;
- **Section 5** contains the conclusions.

1.3. REFERENCES

1.3.1. APPLICABLE DOCUMENTS

The following documents, of the exact issue shown, form part of this document to the extent specified herein. Applicable documents are those referenced in the Contract or approved by the Approval Authority. They are referenced in this document in the form [AD.x]:

Table 1-1 Applicable Documents

Ref.	Title	Code	Version	Date
[AD.1]	Agreement relating to the provision of services pursuant to request for tenders for the provision of GNSS monitoring services	-	-	25/06/19
[AD.2]	19/02 - GNSS Monitoring Services – Contract Extension	-	-	May 22
[AD.3]				
[AD.4]				

1.3.2. REFERENCE DOCUMENTS

The following documents, although not part of this document, amplify or clarify its contents. Reference documents are those not applicable and referenced within this document. They are referenced in this document in the form [RD.x]:

Table 1-2 Reference Documents

Ref.	Title	Code	Version	Date
[RD.1]	Global Positioning System Standard Positioning Service Performance Standard	GPS SPS	5 th Edition	Apr 2020
[RD.2]	Global Positioning System (GPS) Civil Monitoring Performance Specification	DOT-VNTSC-FAA-09-08	-	April 30 th 2009
[RD.3]	Reference Set of Parameters for RAIM Availability Simulations', EUROCAE WG-62	-	-	8-9 July 2003

1.4. ACRONYMS

Acronyms used in this document and needing a definition are included in the following table:

Table 1-3 Acronyms

Acronym	Definition
AOD	Age Of Data
GNSS	Global Navigation Satellite System
GPS	Global Positioning System
HDOP	Horizontal Dilution Of Precision
IAA	Irish Aviation Authority
IGS	International GNSS Service
NANU	Notice Advisory to Navstar Users
NOTAM	Notice To Airmen
PDOP	Position Dilution Of Precision
RAIM	Receiver Autonomous Integrity Monitoring
SIS	Signal In Space
SPS	Standard Positioning Service
TTA	Time To Alarm
UERE	User Equivalent Range Error
URA	User Range Accuracy
URE	User Range Error
VDOP	Vertical Dilution Of Precision

2. INTRODUCTION

2.1. PURPOSE

The purpose of the performance monitoring activity is to collect and analyse data on the performance of the GPS Signal in Space (SIS) [AD.1]. For this report, the applicable requirements are defined in the Global Positioning System Standard Positioning Service Performance Standard (GPS SPS PS), approved by the US Department of Defence [RD.1].

2.2. PERFORMANCE SPECIFICATION AND DEFINITIONS

The applicable performance specifications for the Standard Positioning Service [RD. 1] are as follows, with changes to the previous version of the GPS performance spec (prior to April 2020) noted:

Criteria	Specifications
SPS SIS Accuracy	<p>The User Range Error (URE) for any healthy satellite for Single-Frequency C/A-Code:</p> <ul style="list-style-type: none"> ≤7.0 m 95% Global Average URE during Normal Operations over all age of data (AODs) [previous value was 7.8m] ≤3.8m 95% Global Average URE during Normal Operations at Zero AOD [previous value was 6.0m] ≤9.7 m 95% Global Average URE during Normal Operations at Any AOD [previous value was 12.8m] ≤30 m 99.94% Global Average URE during Normal Operations over one-year period ≤30 m 99.79% Worst Case Single Point Average URE during Normal Operations over one-year period ≤388 m 95% Global Average URE during Extended Operations after 14 Days without Upload. <p>The User Range Error (URE) for all healthy satellites for Single-Frequency C/A-Code:</p> <ul style="list-style-type: none"> ≤2.0 m 95% Global Average URE during Normal Operations over all age of data (AODs) [New specification – did not appear previously] <p>The User Range Rate Error (URRE) for Single-Frequency C/A-Code:</p> <p>≤0.006 m/sec 95% Global Average URRE over any 3-second interval during Normal Operations at Any AOD</p> <p>The User Range Acceleration Error (URAE) for Single-Frequency C/A-Code:</p> <p>≤0.002 m/sec/sec 95% Global Average URAE over any 3-second interval during Normal Operations at Any AOD</p> <p>The UTC Offset Error for Single-Frequency C/A-Code:</p> <p>≤30 nsec 95% Global Average UTCOE during Normal Operations at Any AOD [previous value was 40nsec]</p>
SPS SIS Integrity	<p>The SIS Instantaneous URE Integrity for Single-Frequency C/A-Code:</p>

Criteria	Specifications
	<ul style="list-style-type: none"> • $\leq 1 \times 10^{-5}$ Probability Over Any Hour of the SPS SIS Instantaneous URE Exceeding the NTE Tolerance Without a Timely Alert during Normal Operations <p>The SIS Instantaneous UTCOE Integrity for Single-Frequency C/A-Code:</p> <ul style="list-style-type: none"> • $\leq 1 \times 10^{-5}$ Probability Over Any Hour of the SPS SIS Instantaneous UTCOE Exceeding the NTE Tolerance Without a Timely Alert during Normal Operations <p>The SIS Instantaneous Psat and Pconst for Single-Frequency C/A-Code:</p> <ul style="list-style-type: none"> • $\leq 1 \times 10^{-5}$ Fraction of Time when the SPS SIS Instantaneous URE Exceeds the NTE Tolerance Without a Timely Alert (Psat) [New specification – did not appear previously] • $\leq 1 \times 10^{-8}$ Fraction of Time when the SPS SIS Instantaneous URE from two or more satellites Exceeds the NTE Tolerance due to a common cause Without a Timely Alert (Pconst) [New specification – did not appear previously]
SPS SIS Continuity	<p>SPS SIS Unscheduled Failure Interruption Continuity</p> <ul style="list-style-type: none"> • ≥ 0.9998 Probability Over Any Hour of Not Losing the SPS SIS Availability from a Slot Due to Unscheduled Interruption • Given that the SPS SIS is available from the slot at the start of the hour
Status and Problem reporting	<p>Scheduled Event Affecting Service</p> <ul style="list-style-type: none"> • Appropriate NANU issued to the Coast Guard and the FAA at least 48 hours prior to the event for 95% of the events [previously did not specify a %]
SPS SIS Availability	<p>SPS SIS Per-Slot Availability</p> <ul style="list-style-type: none"> • ≥ 0.957 Probability that a Slot in the Baseline 24-Slot Configuration will be Occupied by a Satellite Broadcasting a Healthy SPS SIS • ≥ 0.957 Probability that a Slot in the Expanded Configuration will be Occupied by a Pair of Satellites Each Broadcasting a Healthy SPS SIS <p>SPS SIS Constellation Availability</p> <ul style="list-style-type: none"> • ≥ 0.98 Probability that at least 21 Slots out of the 24 Slots will be Occupied Either by a Satellite Broadcasting a Healthy SPS SIS in the Baseline 24-Slot Configuration or by a Pair of Satellites Each Broadcasting a Healthy SPS SIS in the Expanded Slot Configuration • ≥ 0.99999 Probability that at least 20 Slots out of the 24 Slots will be occupied either by a Satellite Broadcasting a Healthy SPS SIS in the Baseline 24-Slot Configuration or by a Pair of Satellites Each Broadcasting a Healthy SPS SIS in the Expanded Slot Configuration. • ≥ 0.95 Probability that the Constellation will have at least 24 Operational Satellites regardless of Whether Those Operational Satellites are Located in Slots or Not.
PDOP Availability	<ul style="list-style-type: none"> • $\geq 98\%$ global Position Dilution of Precision (PDOP) of 6 or less • $\geq 88\%$ worst site PDOP of 6 or less
SPS Position Service Availability	<ul style="list-style-type: none"> • $\geq 99\%$ Horizontal Service Availability average location • $\geq 90\%$ Horizontal Service Availability worst-case location

Criteria	Specifications
	<ul style="list-style-type: none"> ≥ 99% Vertical Service Availability average location ≥ 90% Vertical Service Availability worst-case location <p>With 15 m horizontal and 33 m vertical (SIS only) 95% threshold over 24hours <i>[previous values were 17m and 37m]</i></p>
Positioning Accuracy	<ul style="list-style-type: none"> ≤ 8 meters 95% Global Average Horizontal Error <i>[previous value was 9m]</i> ≤ 15 meters 95% worst site Horizontal Error <i>[previous value was 17m]</i> ≤ 13 meters 95% Global Average Vertical Error <i>[previous value was 15m]</i> ≤ 33 meters 95% worst site Vertical Error <i>[previous value was 37m]</i> Global Average Velocity Accuracy ≤ 0.2 m/sec 95% velocity error, any axis <i>[New specification – did not appear previously]</i> ≤ 30 nanoseconds time transfer error 95% of time for Time Transfer Domain Accuracy <i>[previous value was 40nsec]</i>

Table 2-1: SPS Criteria and Specifications

The definitions for each of the criteria and the methodology used for assessment are given below. As well as the GPS SPS [RD.1], the GPS civil monitoring performance specification [RD.2] has also been used to help define the methodology for the assessment.

SPS SIS Accuracy

The SPS SIS accuracy is described in two statistical ways; one way is as the 95th percentile (95%) SPS SIS user range error (URE) at a specified age of data (AOD), the other is as the 95% SPS SIS URE over all AODs. With either statistical expression, the SPS SIS accuracy is also known as the SPS SIS pseudorange accuracy. In this context, “pseudorange” means the full pseudorange data set (i.e., the matched combination of a corrected pseudorange measurement and a pseudorange origin, or equivalently the matched combination of a raw pseudorange measurement and the associated NAV data).

Other accuracy-related SPS SIS performance parameters include the SPS SIS pseudorange rate (velocity) accuracy defined as the 95% SPS SIS pseudorange rate error over all AODs and the SPS SIS pseudorange acceleration (rate rate) accuracy defined as the 95% SPS SIS pseudorange acceleration error over all AODs. These values are not monitored as part of this performance monitoring contract.

SPS SIS Integrity

The SPS SIS integrity is defined as the trust which can be placed in the correctness of the information provided by the SPS SIS. SPS SIS integrity includes the ability of the SPS SIS to provide timely alerts to receivers when the SPS SIS should not be used for positioning or timing. The SPS SIS should not be used when it is providing misleading signal-in-space information (MSI), where the threshold for “misleading” is a not-to-exceed (NTE) tolerance on the SIS URE. For this SPS PS, the four components of integrity are the probability of a major service failure, the time to alert, the SIS URE NTE tolerance, and the alert (either one or the other of two types of alerts).

- Probability of a Major Service Failure. The probability of a major service failure for the SPS SIS is defined to be the probability that the SPS SIS instantaneous URE exceeds the SIS URE NTE tolerance (i.e., MSI) without a timely alert being issued (i.e., unalerted MSI [UMSI]). Alerts generically include both alarms and warnings.

- Time to Alert. The time to alert (TTA) for the SPS SIS is defined to be the time from the onset of MSI until an alert (alarm or warning) indication arrives at the receiver's antenna. Real-time alert information broadcast as part of the NAV message data is defined to arrive at the receiver's antenna at the end of the NAV message subframe which contains that particular piece of real-time alert information.
- SIS URE NTE Tolerance. The SPS SIS URE NTE tolerance for a healthy SPS SIS is defined to be 4.42 times the upper bound on the URA value corresponding to the URA index "N" currently broadcast by the satellite. The SIS URE NTE tolerance for a marginal SPS SIS is not defined and there is no SIS URE NTE tolerance for an unhealthy SPS SIS.

SPS SIS Continuity

The SPS SIS continuity for a healthy SPS SIS is the probability that the SPS SIS will continue to be healthy without unscheduled interruption over a specified time interval. Scheduled interruptions which are announced at least 48 hours in advance do not contribute to a loss of continuity. Scheduled SPS SIS interruptions are announced by way of the Control Segment issuing a "Notice Advisory to Navstar Users" (NANU). NANUs are similar to the "Notices to Airmen" (NOTAMs) issued regarding scheduled interruptions of ground-based air navigation aids. OCS internal procedures are to issue NANUs for scheduled interruptions at least 96 hours in advance.

SPS SIS Availability

The SPS SIS availability is the probability that the slots in the GPS constellation will be occupied by satellites transmitting a trackable and healthy SPS SIS. For this SPS Performance Standard, there are two components of availability as follows:

- Per-Slot Availability. The fraction of time that a slot in the GPS constellation will be occupied by a satellite that is transmitting a trackable and healthy SPS SIS.
- Constellation Availability. The fraction of time that a specified number of slots in the GPS constellation

PDOP Availability

PDOP availability is defined as the percentage of time over a specified time interval that the predicted PDOP is less than a specified value for any point within the service volume [RD.1].

Position Service Availability

Position service availability is defined as the percentage of time over a specified time interval that the position accuracy is less than a specified value for any point within the service volume [RD.1].

Positioning Service Accuracy

Position service accuracy is defined as the statistical difference between position measurements and a surveyed benchmark for any point within the service volume over a specified time interval [RD.1].

2.3. METHODOLOGY

For the performance analysis in this report, raw GPS measurement data from reference stations has been analysed. The primary source of data is the OSi network of active stations in Ireland. OSi operates a national network of GNSS receiver stations. The network consists of 25 receivers that provide 24-hour availability of dual frequency GPS and GLONASS data. For the purposes of this performance monitoring activity, OSi provides raw data for 2 of these sites to GMV NSL for processing and analysis. The sites that are used are Mullingar (MLG1) and Sligo Hospital (SLGO). The locations of these sites are shown in Figure 2-1.

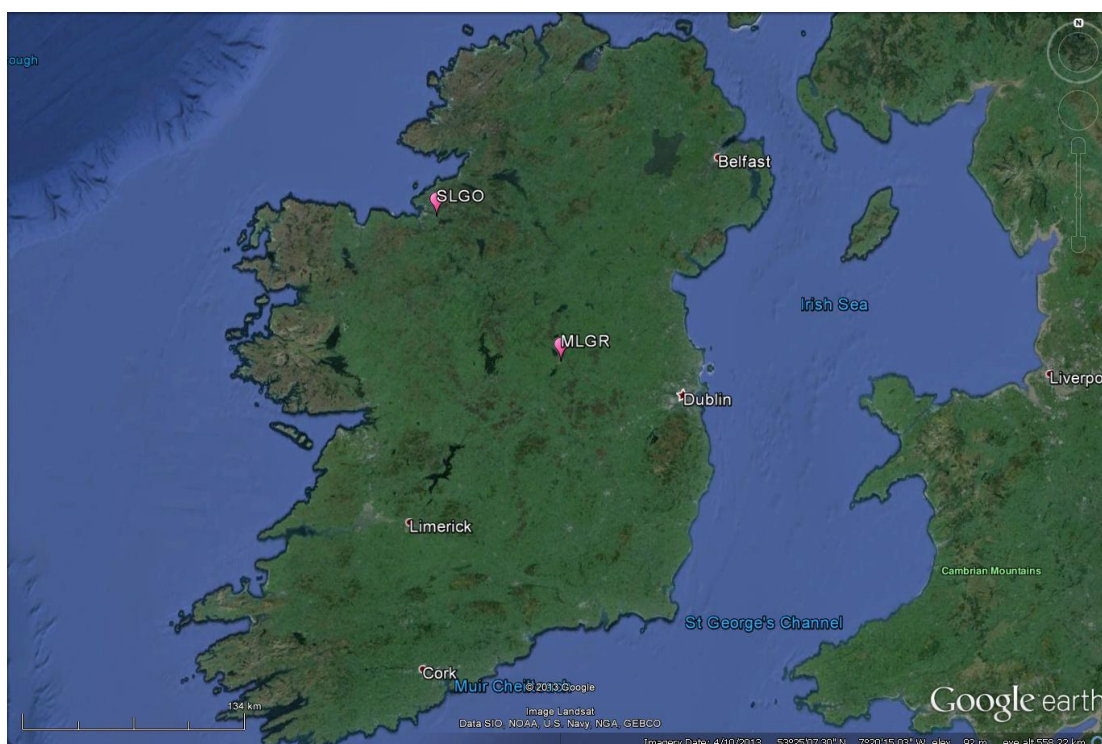


Figure 2-1: Location of Performance Monitoring Stations

In this report, SLGO is used as the main site to provide performance monitoring across the whole of Irish airspace and MLG1 is a back-up in case of problems with SLGO and to allow cross-checking of performance in the case of anomaly investigations. These sites have been chosen to fulfil the following criteria:

- Centrally located within Ireland in order to ensure monitoring of complete airspace;
- Good data availability and continuity (i.e. avoid sites with historically poor data availability);
- Good measurement quality (i.e. avoid sites with known interference, multipath or sky visibility issues);
- Availability of required observation types and data.

In addition to the raw data, NANU information is downloaded from the US Coast Guard Navigation Centre website (<http://www.navcen.uscg.gov/?pageName=gpsNanuInfo>). This provides information on the NANUs for scheduled and unscheduled outages during the monitoring period.

The methods for assessing of each of the requirements are described below.

SPS SIS Accuracy

SIS accuracy is assessed through processing and analysis of the raw measurement data. In order to compute the SIS accuracy, the measurements recorded at the GPS receiver are used to compute the instantaneous SIS errors. This is done by computing the difference between computed ranges (based on known receiver location and satellite position) and the corrected measurement, which has satellite and receiver clock biases, group delay, ionospheric and tropospheric errors removed. Once the SIS range errors for every satellite measurement on every epoch have been computed, the per-satellite and all satellite statistics across the whole period, as well as daily statistics for all satellites combined, are generated.

SPS SIS Integrity

SIS accuracy is assessed through processing and analysis of the raw measurement data. The SIS integrity is assessed by comparing each instantaneous computed SIS error value with a threshold value of $4.42 \times$ broadcast URA. The number of occasions where the instantaneous URE exceeds the threshold are counted and checked against the expected number of failures.

SPS SIS Continuity

SIS continuity is assessed through analysis of the broadcast navigation messages and the NANU archive. Firstly, the daily broadcast navigation messages are scanned in order to find the time periods for any satellites that do not have healthy navigation messages. These satellites and time periods are then matched against NANU information to see if the outages are scheduled or unscheduled.

The SIS continuity is computed for the baseline 24-slot constellation and is an average value over all slots. The total time that any satellites in the baseline constellation were unhealthy due to an unscheduled outage is divided by the total time in the analysis period and expressed as a percentage. Results are presented for the reporting period and, when available, for the previous year.

SPS SIS Availability

SIS availability is assessed through analysis of the broadcast navigation messages and the NANU archive. Firstly, the daily broadcast navigation messages are scanned in order to find the time periods for any satellites that do not have healthy navigation messages. These satellites and time periods are then matched against NANU information to see if the outages are scheduled or unscheduled.

The SIS availability is computed for the baseline 24-slot constellation as well as for the whole constellation and is an average value over all slots. At each epoch the number of healthy satellites (both in the baseline 24-slot constellation and in total) is counted. Then the following parameters are computed:

- Total time that there are less than 21 healthy satellites in the baseline constellation;
- Total time that there are less than 20 healthy satellites in the baseline constellation;
- Total time that there are less than 24 healthy satellites in the whole constellation.

These parameters are then divided by total time of the analysis and expressed as percentage values. Results are presented for the reporting period and, when available, for the previous year.

It should be noted that in case the baseline 24-slot constellation does not meet requirements, the analysis will be expanded to include pairs of satellites in the expanded slot constellation.

PDOP Availability

PDOP availability is assessed through processing and analysis of the raw measurement data. The PDOP availability is assessed by computing the PDOP for all satellites in view above 5 degrees at the GPS receiver at every epoch (1Hz rate). Each PDOP value is checked against the threshold value of 6 and any failures are counted. The numbers of failures on each day are then used to generate the daily availability value. A separate availability value for each day is computed.

Position Service Availability

Position service availability is assessed through processing and analysis of the raw measurement data. The derivation of the position service availability requirements of 15m (95% horizontal accuracy) and 33m (95% vertical accuracy) for 99% of the time are explained a bit more in section B.3.2 of the GPS SPS [RD.1]. The requirement is based on fulfilling a 1-sigma UERE of 3.6m, HDOP of 2.1 and VDOP of 4.53. To check this requirement, the following approach is used:

- For each day, compute daily rms SIS error for all satellites combined. This is equivalent to the 1-sigma UERE in the description above;

- On each epoch, multiply daily rms SIS error by HDOP value to compute estimated horizontal accuracy due to SIS error;
- For each epoch, multiply daily rms SIS error by VDOP value to compute estimated vertical accuracy due to SIS error;
- Compute daily availability (%) of estimated horizontal accuracy < 7.5m (1-sigma);
- Compute daily availability (%) of estimated vertical accuracy < 16.5m (1-sigma).
- If daily availability of horizontal accuracy greater than the required threshold, the requirement for horizontal service accuracy is passed;
- If daily availability of vertical accuracy greater than the required threshold, the requirement for vertical service accuracy is passed.

Positioning Service Accuracy

In order to check the position service accuracy, the raw measurements recorded at the GPS receiver are used to compute a user position solution on every epoch (1Hz). The computed positions are then compared against the known position of the receiver in order to generate horizontal and vertical position errors. Statistics for 95% error value, 99.99% error value etc. are then computed separately for each day and checked against the thresholds.

2.4. ASSUMPTIONS

For processing the raw data and generating the results the following assumptions are made:

- Single frequency (L1) processing with C/A code;
- 5-degree elevation mask used;
- Broadcast iono model (Klobuchar) used to remove ionospheric errors;
- RTCA trop model used to remove tropospheric errors;
- Weighted least squares RAIM algorithm used for RAIM prediction (protection level computation) and Fault Detection;
- Probability of missed detection = 0.001 and Probability of false alarm = 1×10^{-5} for RAIM computations;
- UERE budget (non-SIS components) used in position solution and for RAIM predictions based given below [RD.3]:

Elevation, degrees	Error, metres
5	7.48
10	6.64
15	5.92
20	5.31
30	4.31
40	3.57
50	3.06
60	2.73
90	2.44

- The URA value from the broadcast navigation message is combined with the values in the table to form the total UERE for the observations.

As the actual monitoring is based on the measurements from one receiver, the following points should be noted:

- Performance monitoring is local to the monitoring station with a coverage area defined by the correlation of the major error sources and the configuration of the constellation.
- The range domain errors contain the residuals of other error sources other than the SIS range errors, hence the performance statistics generated are conservative.

3. SPS PERFORMANCE

3.1. BASELINE 24-SLOT CONSTELLATION

The SPS SIS performance standard is largely based on the GPS baseline 24-slot constellation, which consists of 24 slots in six orbital planes with four slots per plane. Some of these slots are expanded, whereby two satellites occupy fore and aft positions at that slot, in which case the slot is occupied as long as at least one of the expanded slots is occupied by an operational satellite. It is important to identify the baseline constellation (and expanded slots) to act as reference to subsequent data processing and analysis. The following tables show the satellite PRN in each slot for the baseline constellation for the period July to September 2023¹.

Table 3-1: Baseline constellation in the Period 1 July to 30 September 2023

Slot	A1	A2A/A2F	A3	A4	B1A/B1F	B2	B3	B4	C1	C2	C3	C4A/C4F
PRN	24	31/28	30	7	16/26	25	14	12	29	27	8	19/17
Slot	D1	D2A/D2F	D3	D4	E1	E2	E3A/E3F	E4	F1	F2A/F2F	F3	F4
PRN	11	1/21	18	6	3	10	5/20	23	32	15/13	9	4

It is noted that during this period, SVN63 (PRN1) from slot D2A had an issue on 10th July and was set unhealthy, and then was decommissioned on 10th August. Another satellite (SVN44) was re-activated to broadcast PRN22 from 18th August but there is no information available on the plane and slot in which that satellite is located.

3.2. SPS SIS ACCURACY

In addition to the specifications in Table 2-1, the Conditions and Constraints for SPS SIS URE Accuracy specification [RD.1] are:

- For any healthy SPS SIS
- Neglecting single-frequency ionospheric delay model errors
- Including group delay time correction (TGD) errors at L1
- Including inter-signal bias (P(Y)-code to C/A-code) errors at L1

The statistics presented here are based on the same sample rate for positioning (1Hz). It should be noted that the computed range errors (in addition to SIS errors) contain residual errors local to the monitoring antenna (multipath, tropospheric and ionospheric). The URE Accuracy (95th percentile) values of each satellite for the period July to September 2023 are shown in the next figure.

¹ The information on slots is taken from the figure at <https://www.navcen.uscg.gov/pdf/gps/current.pdf>. It is noted that there is some inconsistency between this figure and the slot numbers in the ops advisory messages. The figure was last updated on 1st March 2023 and before that on 1st June 2022.

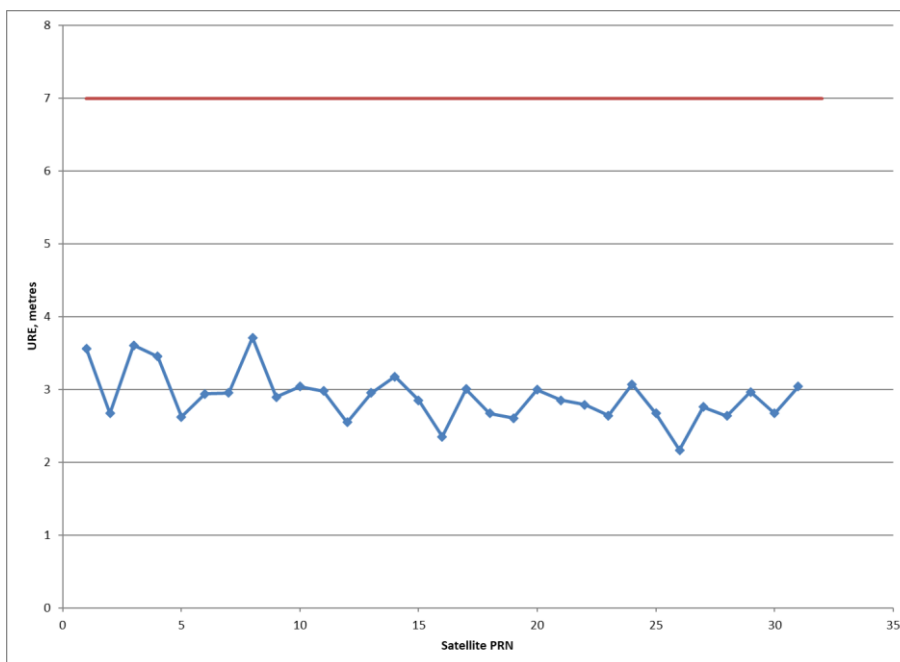


Figure 3-1: Constellation URE (95%) for Reporting Period

It can be seen that the URE (95%) for all satellites is well below the 7m threshold.

The daily constellation RMS URE results in the period July to September 2023 and the 3.6m threshold are shown in the next figure. Note that ≤ 7 m 95% SPS SIS URE performance standard is equivalent to a ≤ 3.6 m RMS SPS SIS URE performance standard [RD.1]. This is also important for the position service availability assessment.

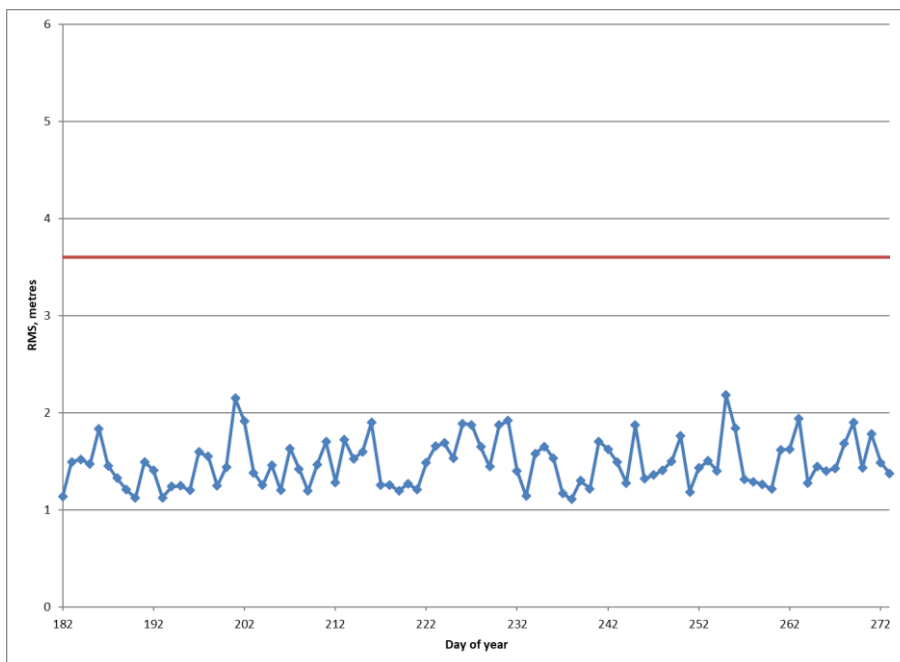


Figure 3-2: Constellation RMS URE for Reporting Period

It can be seen that the RMS values are below the threshold (3.6 metres) on all days.

As well as the 95% and rms URE statistics, additional URE statistics are computed, including mean, 1-sigma and maximum values. Although not strictly required for the performance specification, these values can be useful for anomaly investigation. The range error statistics (in metres) for the period July to September 2023 are given in the table below.

Table 3-2: Range Error Statistics for Reporting Period

PRN	Range Error (mean)	Range Error (RMS)	1-sigma	Range Error (95%)	Range Error (max)	Number of Samples
1	-0.98	1.93	1.66	3.56	32.08	246115
2	0.03	1.41	1.41	2.67	11.25	2447755
3	-0.73	1.89	1.74	3.61	8.33	2585808
4	-1.19	1.79	1.33	3.46	6.40	2795938
5	-0.32	1.38	1.34	2.63	6.19	2726945
6	-0.80	1.63	1.42	2.94	6.13	2796221
7	-0.84	1.58	1.34	2.95	9.06	2896496
8	0.07	1.78	1.78	3.71	8.28	2505904
9	-0.60	1.49	1.37	2.90	7.69	2635597
10	-0.45	1.53	1.47	3.04	6.58	2923837
11	-0.92	1.62	1.33	2.98	6.48	2856043
12	-0.10	1.39	1.39	2.55	8.89	2612521
13	-0.53	1.53	1.43	2.96	9.37	2386094
14	-0.94	1.66	1.37	3.18	9.45	2894588
15	-0.33	1.41	1.38	2.85	5.04	2415948
16	0.27	1.17	1.13	2.35	7.12	2604729
17	-0.86	1.67	1.43	3.01	6.93	2888619
18	-0.23	1.35	1.34	2.67	6.90	2849181
19	0.04	1.32	1.32	2.61	9.63	2870742
20	0.21	1.53	1.51	3.00	7.12	2667394
21	0.35	1.50	1.46	2.85	16.82	2438346
22	-0.33	1.46	1.43	2.79	7.39	1374214
23	-0.47	1.34	1.26	2.64	7.30	2910063
24	-0.67	1.63	1.49	3.07	5.59	2120915
25	0.39	1.27	1.21	2.67	6.98	2321108
26	0.29	1.11	1.07	2.17	4.77	2468863
27	0.19	1.44	1.43	2.76	8.15	2351684
28	-0.50	1.36	1.27	2.64	5.67	2071923
29	0.37	1.58	1.53	2.97	8.31	2751538
30	-0.44	1.45	1.38	2.67	9.39	2749732
31	-0.62	1.59	1.47	3.04	7.52	2796373
32	-0.14	1.31	1.30	2.46	7.40	2890715
All	-0.33	1.32	1.28	2.92	32.08	80851949

Overall, the measured SIS accuracy for any satellite is below the threshold values throughout the monitoring period for all satellites.

The measured accuracy for all satellites combined is above the threshold of 2m, which is likely due to ionospheric residual errors affecting the analysis.

3.3. SPS SIS INTEGRITY

In addition to the specifications in Table 2-1, the Conditions and Constraints for SPS SIS Integrity performance [RD.1] are:

For any healthy SPS SIS;

SPS SIS URE NTE tolerance defined to be ± 4.42 times the upper bound on the URA value corresponding to the URA index "N" currently broadcast by the satellite;

Given that the maximum SPS SIS instantaneous URE did not exceed the NTE tolerance at the start of the hour;

Worst case for delayed alert is 6 hours;

Neglecting single-frequency ionospheric delay model errors.

Based on the requirement of 1×10^{-5} /hr probability for misleading information, 92-day period and a 31-satellite constellation, the maximum number of events expected is 0.66.

On every epoch throughout the monitoring period, the instantaneous measured URE for each satellite has been compared against a threshold of 4.42 times the upper value of the URA index. The number of URE values above the threshold has been recorded and is checked against the expected number.

From the analysis there are 4 events on different days where this condition is met, which could indicate a failure:

- On 4th and 8th September, there are short events (a few seconds) for one satellite at a time. Both of these are for quite low elevation satellites that have low SNR and the issue is transient, indicating it is likely to be difficulties with tracking or multipath. In addition, the satellites affected do not indicate a problem at other sites at the same time;
- On 10th July there are SIS failures lasting for more than 20 minutes on PRN1. The range error is around 30m and for most of this period is detected by the RAIM Fault Detection algorithm implemented in the positioning engine, although there are 117 seconds before detection when the range error is large and exceed the tolerance. This appears to be a real fault related to NANU 2023036 and is analysed in more detailed in Annex A;
- On 22nd August there are SIS failures lasting for 16 seconds on PRN15. The range error is around 30m and for all of this period is detected by the RAIM Fault Detection algorithm implemented in the positioning engine. This appears to be a real fault related to NANU 2023047 and is analysed in more detailed in Annex A.

There do appear to be two real SIS failures, but they are alerted within the 6hr limit and therefore pass.

3.4. SPS SIS CONTINUITY

In addition to the specifications in Table 2-1, the Conditions and Constraints for SPS SIS Continuity performance [RD.1] are:

- Calculated as an average over all slots in the 24-slot constellation, normalized annually;
- Given that the SPS SIS is available from the slot at the start of the hour.

During this reporting period there were 5 unscheduled events affecting the baseline constellation, lasting for a total of 1313.22 hrs. Therefore this gives a continuity figure of 97.52% in this period, which does meet the requirement of 99.98% in this period.

3.5. SPS SIS AVAILABILITY

In addition to the specifications in Table 2-1, the Conditions and Constraints for SPS SIS Availability performance [RD.1] are:

Calculated as an average over all slots in the 24-slot constellation, normalized annually;

Applies to satellites broadcasting a healthy SPS SIS which also satisfy the other performance standards in this SPS Performance Standard.

The total period (in this monitoring period) in which satellites from the baseline 24-satellite constellation broadcast an unhealthy SIS was 1340.88 hours². This is equivalent to an average of 0.9747 over all slots in the 24-slot constellation and satisfies SPS SIS Per-slot Availability standard (≥ 0.957).

The minimum number of the baseline constellation satellites broadcasting healthy SPS SIS was 22, greater than the specifications of 20 and 21. Hence, performance during the monitoring period was measured at the 100% level, satisfying the Performance Standard as specified below.

- ≥ 0.98 Probability that at least 21 Slots out of the 24 Slots will be Occupied Either by a Satellite Broadcasting a Healthy SPS SIS in the Baseline 24-Slot Configuration or by a Pair of Satellites Each Broadcasting a Healthy SPS SIS in the Expanded Slot Configuration;
- ≥ 0.99999 Probability that at least 20 Slots out of the 24 Slots will be occupied either by a Satellite Broadcasting a Healthy SPS SIS in the Baseline 24-Slot Configuration or by a Pair of Satellites Each Broadcasting a Healthy SPS SIS in the Expanded Slot Configuration.

The minimum number of operational satellites broadcasting healthy messages in this reporting period was 29. This represents performance at the 100% level, satisfying the Performance Standard as specified below.

- ≥ 0.95 Probability that the Constellation has at least 24 operational satellites regardless of whether the operational satellites are located in the baseline slots.

3.6. PDOP AVAILABILITY

In addition to the specifications in Table 2-1, the Conditions and Constraints for PDOP performance [RD.1] are:

Defined for position solution meeting the representative user conditions and operating within the service volume over any 24-hour interval;

Based on using only satellites transmitting standard code and indicating "healthy" in the broadcast navigation message.

The following plot shows the daily PDOP availability ($PDOP < 6$) calculated at the site for all healthy satellites above 5 degrees elevation during the period July to September 2023.

² These figures are computed assuming that since SVN63 / PRN1 was decommissioned, it no longer formed part of the baseline constellation.

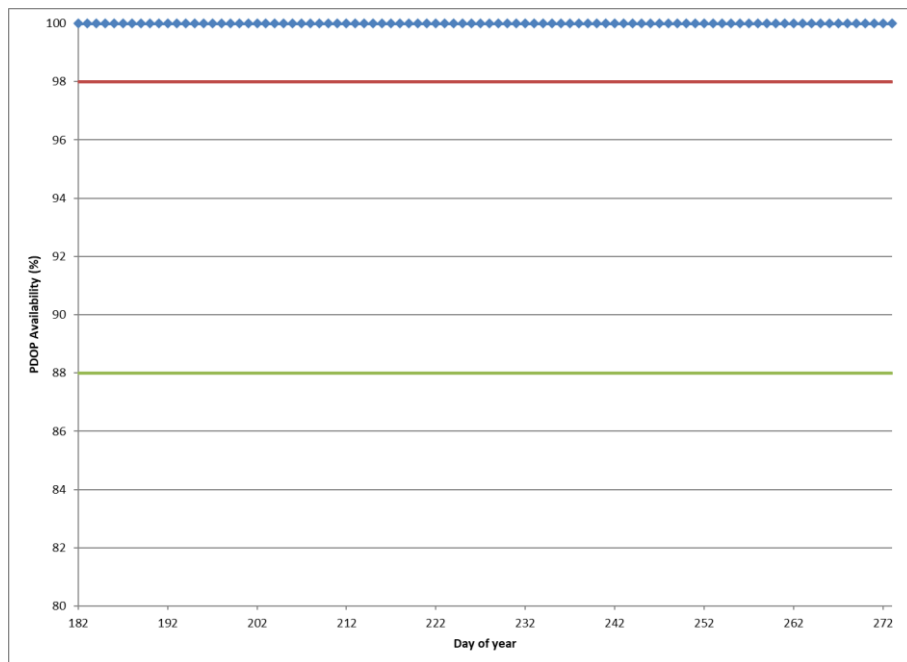


Figure 3-3: Daily PDOP Availability in the Reporting Period

It can be seen that the daily PDOP availability values are all above the thresholds of 98% (global average) and 88% (worst site). Therefore, the PDOP availability fulfils the requirements.

In addition, the daily mean and maximum PDOP values are displayed for the same period.

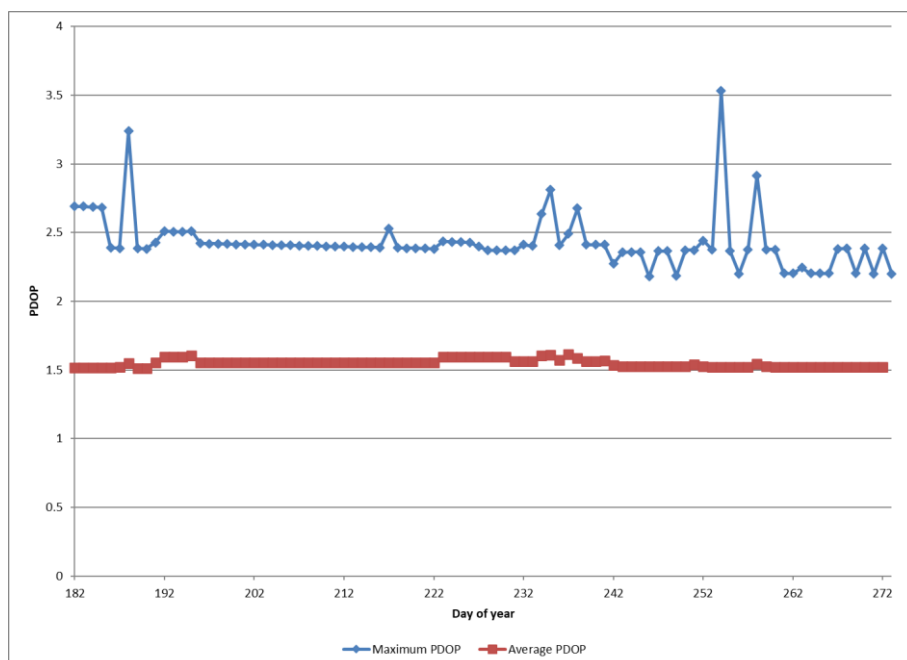


Figure 3-4: Daily Maximum PDOP Value in the Reporting Period

The daily PDOP values PDOP can be used to identify specific days that have different performance from the others. It can be seen that on all days the maximum value is much less than 6.

3.7. POSITION SERVICE AVAILABILITY

In addition to the specifications in Table 2-1, the Conditions and Constraints for Service Availability performance [RD.1] are:

15 meters horizontal (SIS only) 95% threshold;

33 meters vertical (SIS only) 95% threshold;

Defined for position solution meeting representative user conditions and operating within the service volume over any 24-hour interval;

Based on using only satellites transmitting standard code and indicating "healthy" in the broadcast navigation message.

The computation of these values is detailed in section 2.3.

The daily horizontal and vertical service availabilities for the period July to September 2023 are shown in the following figures.

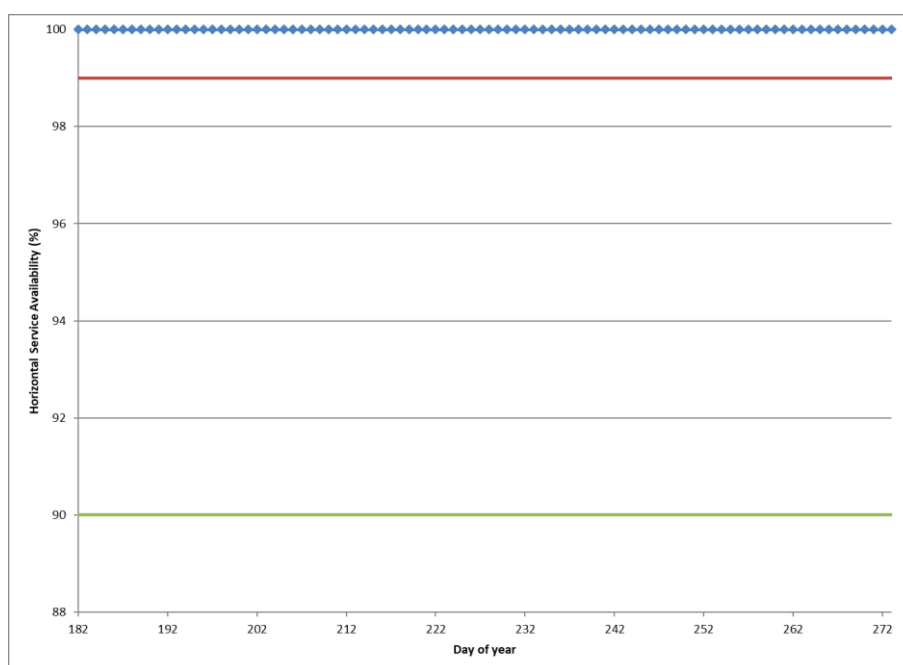


Figure 3-5: Daily Horizontal Service Availability Values for Reporting Period

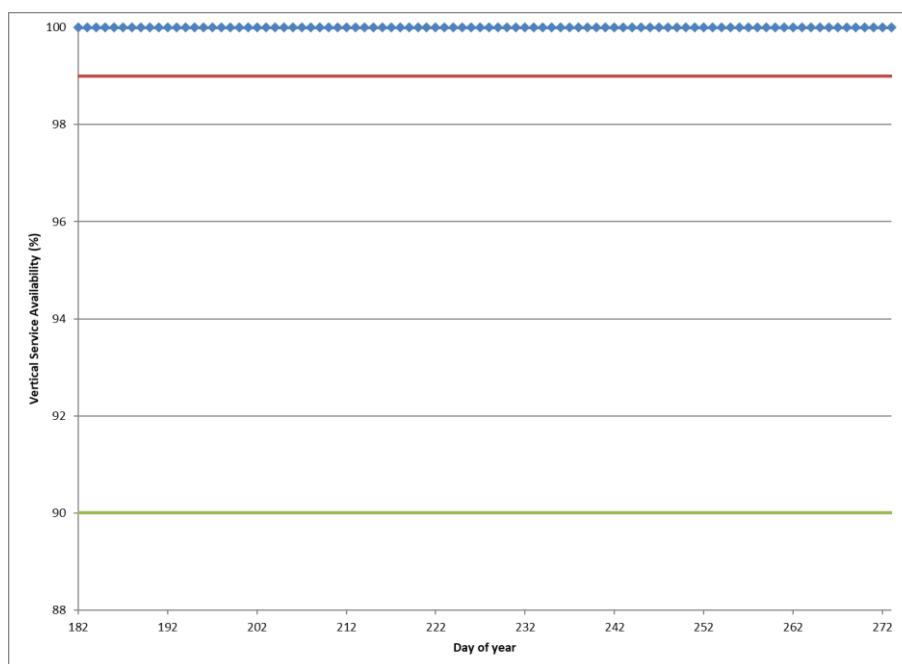


Figure 3-6: Daily Vertical Service Availability Values for Reporting Period

These plots show the horizontal and vertical availability are well above the thresholds of 99% (global average) and 90% (worst site) for the reporting period. Therefore, the position service availability fulfils the requirements.

3.8. POSITIONING ACCURACY

In addition to the specifications in Table 2-1, the Conditions and Constraints for Positioning Accuracy performance [RD.1] are:

Defined for position solution meeting the representative user conditions;

Standard based on a measurement interval of 24 hours averaged over all points within the service volume.

For this monitoring activity it should be noted that the position accuracy is assessed through analysis of real data at a single point, rather than through service volume analysis.

The daily horizontal and vertical accuracy values (95%) for the period July to September 2023 are shown in the following figures.

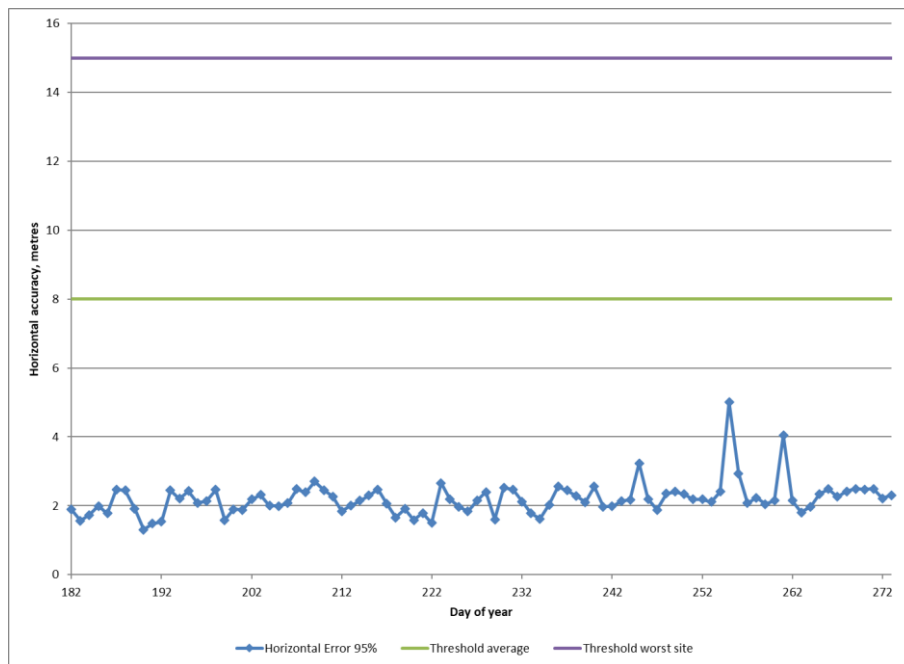


Figure 3-7: Daily Horizontal Position Accuracy (95%) for Reporting Period

It can be seen that the daily horizontal accuracy values are all below the thresholds of 8m (global average) and 15m (worst site).

Also, the daily vertical accuracy values are well below the thresholds of 13m (global average) and 33m (worst site).

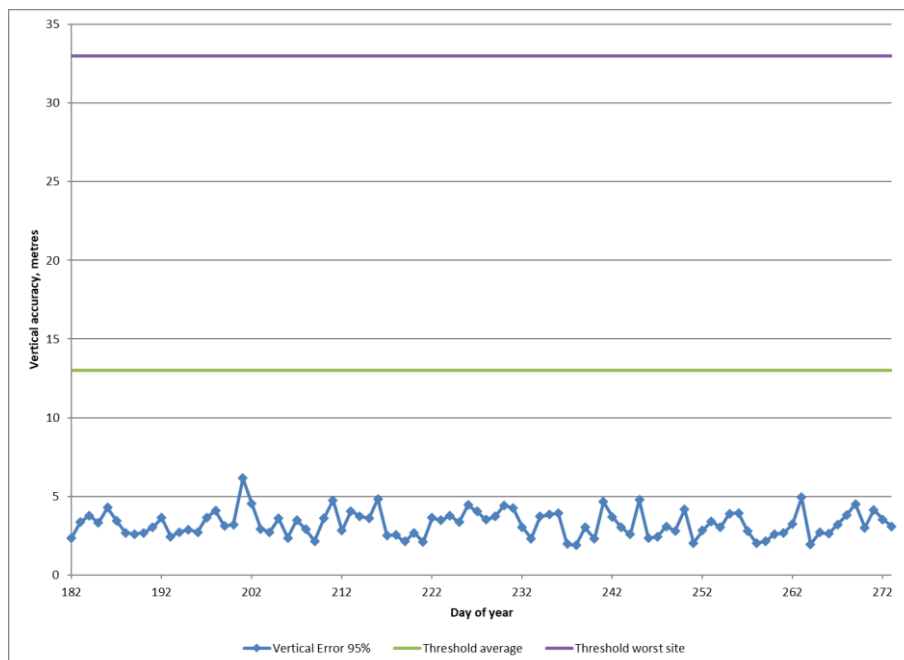


Figure 3-8: Daily Vertical Position Accuracy (95%) for Reporting Period

In addition, the daily position accuracy values at the 99.99% level are shown for the same period.

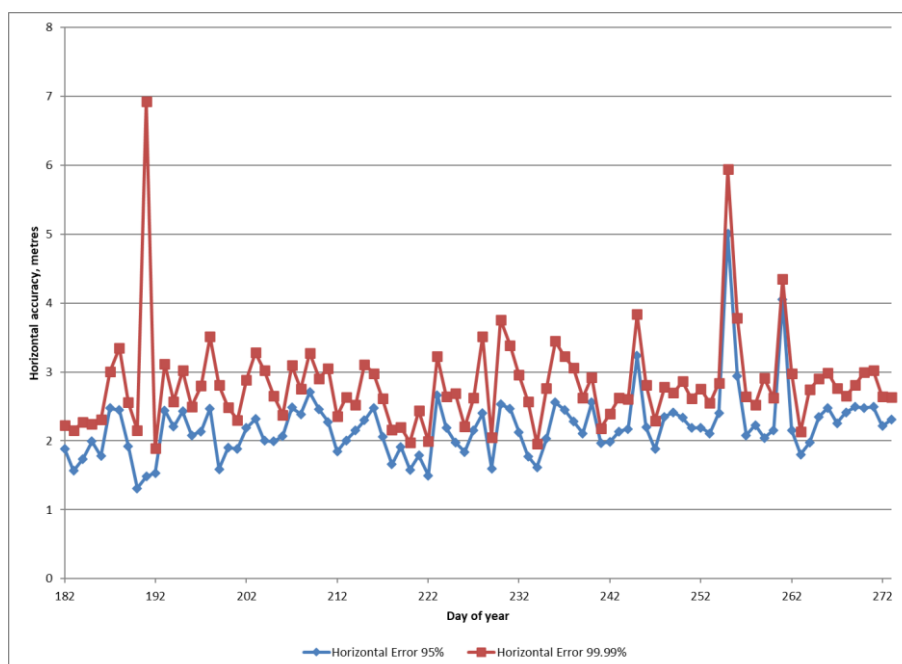


Figure 3-9: Daily Horizontal Position Accuracy (99.99%) for Reporting Period

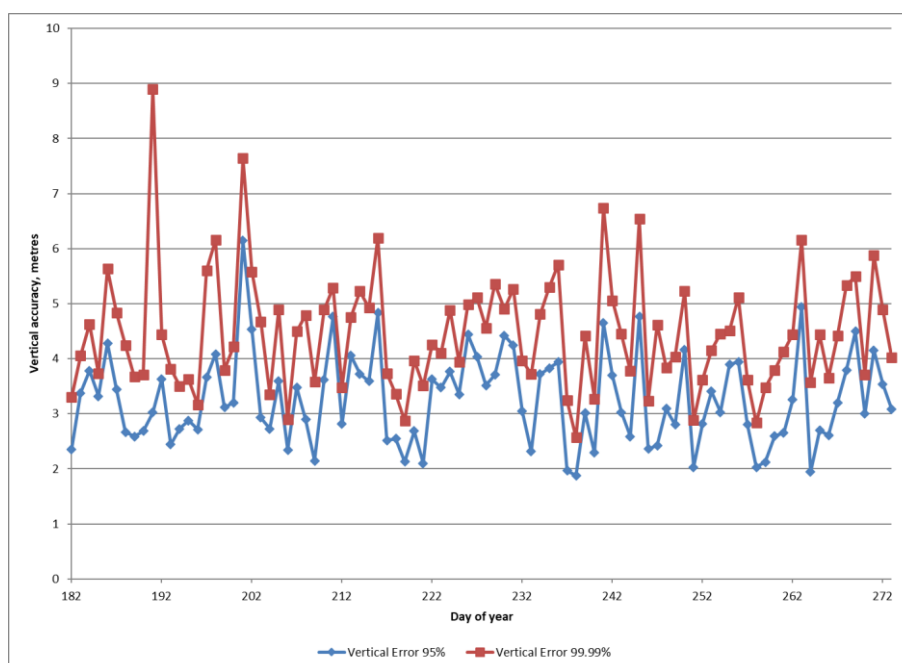


Figure 3-10: Daily Vertical Position Accuracy (99.99%) for Reporting Period

It can be seen that the 99.99% values generally follow the same pattern as the 95% values and are not significantly larger.

4. NANU ANALYSIS

(<http://www.navcen.uscg.gov/?pageName=gpsNanuInfo>). Summaries of the forecast and actual outages for scheduled and unscheduled events are given below. NANUs that affect a whole slot in the baseline constellation are highlighted in green. NANUs that affect one satellite of an expanded slot in the baseline constellation are highlighted in blue, tan, orange, red or purple.

Table 4-1: Summary of Forecast Scheduled Outages

NANU	PRN	Type	Start day	Start Time	Stop day	Stop time	Outage (hours)	Ref
2023033	17	FCSTDV	187	2115	188	915	12	C4F
2023035	30	FCSTDV	195	145	195	1345	12	A3
2023045	32	FCSTDV	237	230	237	1430	12	F1
2023050	12	FCSTDV	244	130	244	1330	12	B4
2023053	2	FCSTDV	250	1945	251	745	12	D5
2023055	9	FCSTDV	257	2345	258	1145	12	F3

Table 4-2: Summary of Actual Scheduled Outages

NANU	PRN	Type	Start day	Start Time	Stop day	Stop time	Outage (hours)	Ref
2023034	17	FCSTSUMM	187	2117	188	339	6.3666	2023033
2023041	30	FCSTSUMM	195	148	195	659	5.1833	2023035
2023049	32	FCSTSUMM	237	234	237	740	5.1000	2023045
2023052	12	FCSTSUMM	244	137	244	726	5.8167	2023050
2023054	2	FCSTSUMM	250	2023	251	225	6.0333	2023053
2023056	9	FCSTSUMM	257	2352	258	504	5.2000	2023055

Table 4-3: Summary of Cancelled Outages

NANU	PRN	Type	Start day	Start Time	Stop day	Stop time	Ref
-	-	-	-	-	-	-	-

Table 4-4: Summary of Forecast and Actual Unscheduled Outages

NANU	PRN	Type	Start day	Start Time	Stop day	Stop time	Outage (hours)	Ref
2023036	1	UNUSUFN	191	1020	NA	NA	NA	D2A
2023044	1	DECOM	191	1020	222	2200	755.6666	2023036
2023037	28	UNUSUFN	191	1110	NA	NA	NA	A2F
2023038	28	UNUSABLE	191	1110	191	1649	5.65	2023037
2023039	28	UNUSUFN	191	2300	NA	NA	NA	A2F
2023040	28	UNUSABLE	191	2300	195	107	74.1166	2023039
2023043	28	UNUSUFN	222	2352	NA	NA	NA	A2F

NANU	PRN	Type	Start day	Start Time	Stop day	Stop time	Outage (hours)	Ref
2023051	28	UNUSABLE	222	2352	241	2304	455.2	2023043
2023047	15	UNUSUFN	234	1611	NA	NA	NA	F2A
2023048	15	UNUSABLE	234	1611	235	1446	22.5833	2023047

The constellation availability and continuity figures for the baseline constellation, and for all satellites, based on the NANU information are shown in the following table. Note that for continuity and availability, the baseline constellation is affected if at least one of the satellites in an expanded slot is healthy, i.e. an outage on one of the satellites in an expanded slot still affects the statistics for the baseline constellation.

Table 4-5: Summary of NANU Statistics for Monitoring Period

	Q3 2023
Hrs	2208
total forecast downtime (all)	72.00
total forecast downtime (baseline)	60.00
total actual scheduled downtime (all)	33.70
total actual scheduled downtime (baseline)	27.67
Scheduled satellite outage events (all)	6
Scheduled satellite outage events (baseline)	5
Unscheduled satellite outage events (all)	5
Unscheduled satellite outage events (baseline)	5
Total actual unscheduled downtime (all)	1313.22
Total actual unscheduled downtime (baseline)	1313.22
Total actual downtime (all)	1346.92
Total actual downtime (baseline)	1340.88
Availability (all)	98.032
Availability (baseline)	97.470
Continuity (baseline)	97.520

5. CONCLUSIONS

The following table summarises the measured performance against the specification.

Table 5-1: Summary of Performance

Criteria	Specifications	Measured Performance	Passed
SPS SIS Accuracy	The User Range Error (URE) ≤ 7 m 95% for any satellite	All SV < 7 m	Yes
	The User Range Error (URE) ≤ 2 m 95% for all satellites	2.9m	No. Although likely due to ionospheric residuals
SPS SIS rms	≤ 3.6 m	All days < 3.6 m	Yes
SPS SIS Integrity	The SIS Integrity $\leq 1 \times 10^{-5}$ Probability Over Any Hour (< 0.7 events per quarter)	No events	Yes Two possible events but alerts raised within 6 hrs.
SPS SIS Continuity	≥ 0.9998 Probability Over Any Hour	97.52% (5 outages on baseline)	No
SPS SIS Availability	SPS SIS Per-Slot Availability <ul style="list-style-type: none"> ≥ 0.957 SPS SIS Constellation Availability <ul style="list-style-type: none"> ≥ 0.98 Probability that at least 21 Slots out of the 24 Slots will be healthy ≥ 0.99999 Probability that at least 20 Slots out of the 24 Slots will be healthy ≥ 0.95 Probability that the Constellation will have at least 24 Operational Satellites 	1) 97.5% per-Slot Availability 2) 100% Constellation Availability 3) 100% probability that the number of operational satellites is larger than 24.	Yes
PDOP Availability	<ul style="list-style-type: none"> $\geq 98\%$ global PDOP of 6 or less $\geq 88\%$ worst site PDOP of 6 or less 	$> 99.8\%$ availability on all days	Yes

Criteria	Specifications	Measured Performance	Passed
SPS Position Service Availability	<ul style="list-style-type: none"> • $\geq 99\%$ Horizontal Service Availability average location • $\geq 90\%$ Horizontal Service Availability worst-case location • $\geq 99\%$ Vertical Service Availability average location • $\geq 90\%$ Vertical Service Availability worst-case location 	100% availability on all days	Yes
Positioning Accuracy	<ul style="list-style-type: none"> • ≤ 8 meters 95% All-in-View Global Average Horizontal Error (SIS Only) • ≤ 15 meters 95% All-in-View worst site Horizontal Error (SIS Only) • ≤ 13 meters 95% All-in-View Global Average Vertical Error (SIS Only) • ≤ 33 meters 95% All-in-View worst site Vertical Error (SIS Only) 	1) < 5 metres 95% Horizontal Error at the site 2) < 6 metres 95% Vertical Error at the site	Yes

From the table it can be seen that the measured performance is within the required values for almost all requirements. The exception is the SIS accuracy for all SVs, which is slightly larger than the requirement but is likely due to the computation method including ionospheric residual errors and the SIS continuity, which fails due to several long unplanned outages in this quarter.

ANALYSIS OF SIS INTEGRITY EVENTS

This annex contains some analysis of SIS integrity events identified in the current monitoring period.

10TH JULY 2023

On 10th July 2023, the range residual errors measured on satellite SVN63 / PRN01 exceed the threshold ($4.42 \times \text{URA}$) for over 20 minutes. For some of this time, RAIM FD flags there is a fault but there are some epochs with high range error, and some increase in position error.

The first plot shows the range residual error (as measured at SLGO) for PRN01 on 10th July. It can be seen that at first it is small but suddenly jumps to more than 30m just before 09:57. It remains at this high level until the satellite is stopped being used.

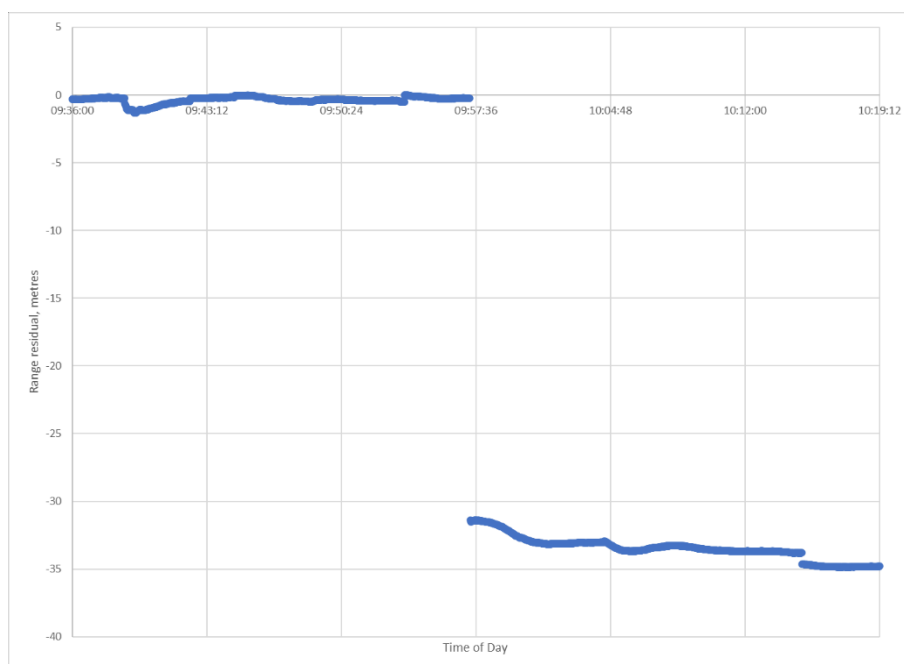


Figure 5-1: Range Residual Error on PRN01 on 10th July 2023 at SLGO

This increase in range error for PRN1 affects the position solution (both horizontal and vertical) so that the error is larger than usual during this period, although not excessive. This is shown in the next figure.



Figure 5-2: Position Error during period of High PRN01 Error on 10th July 2023 at SLGO

Finally, we can see that it is possible to detect this error though RAIM FD. In this figure the weighted least squares RAIM test statistic (which is a measure of the consistency of the residuals) is shown, along with the detection threshold. It can be seen that after 09:58 the test statistic grows larger than the threshold, indicating a fault in one of the measurements.

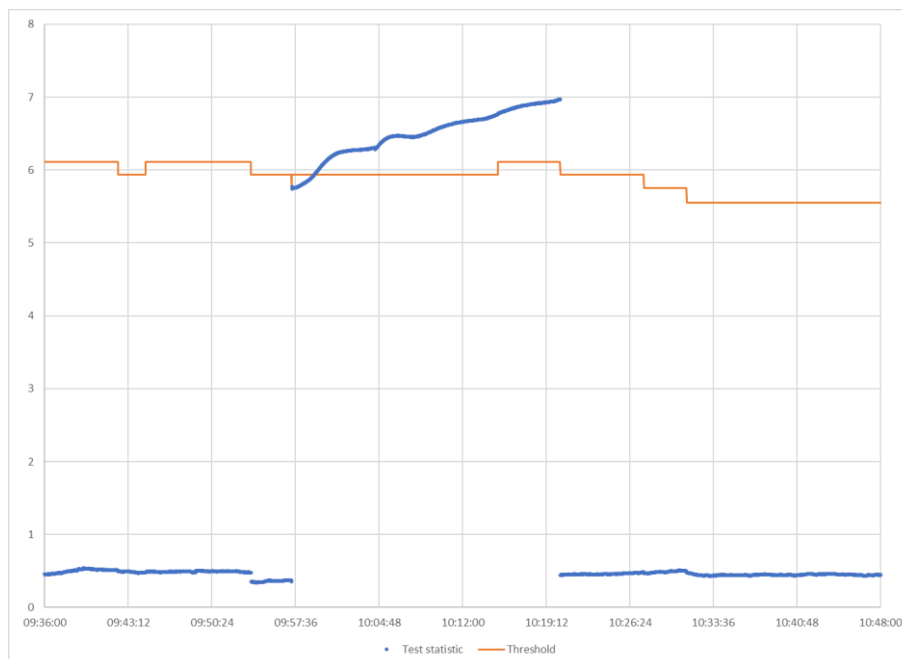
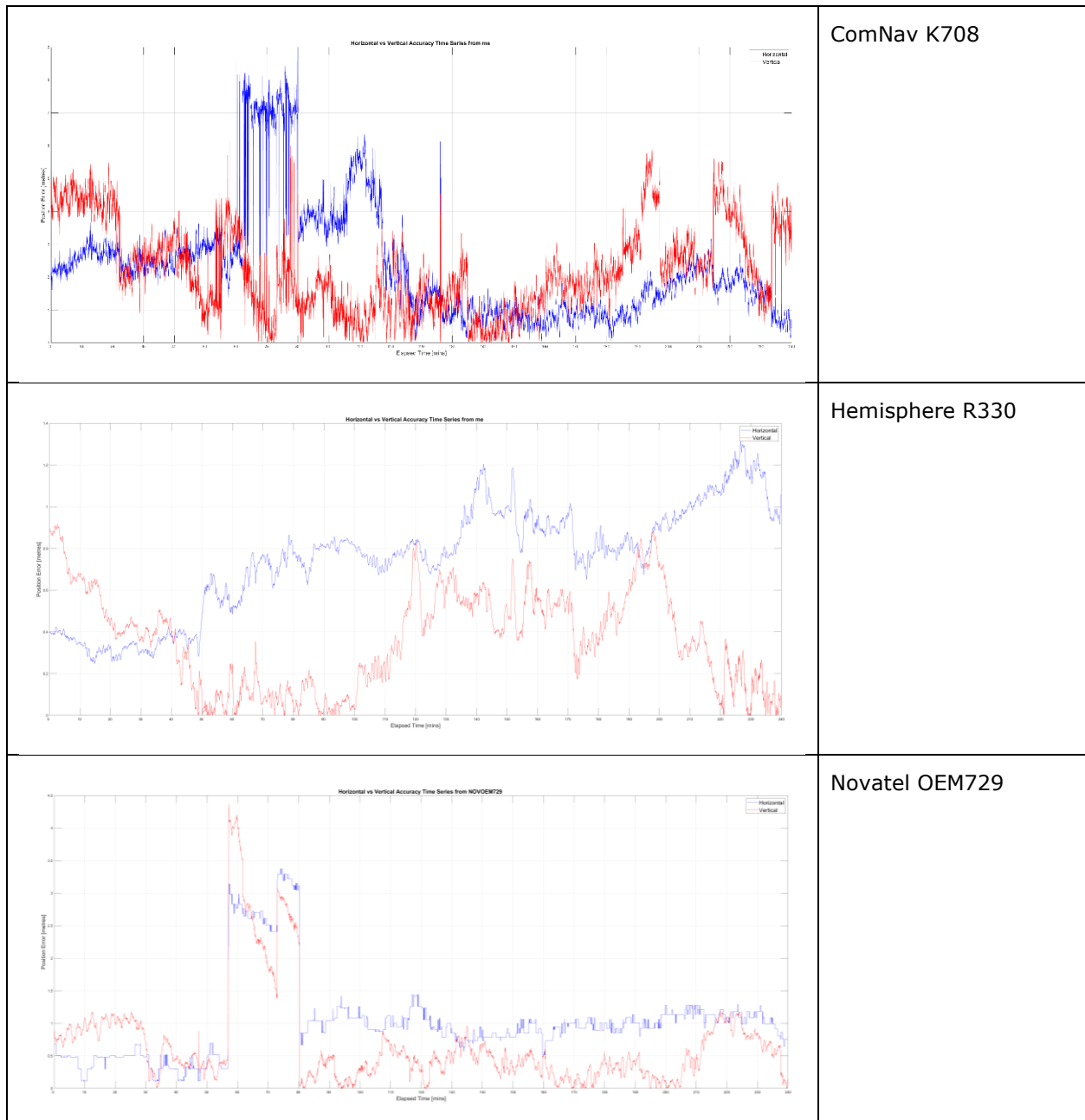


Figure 5-3: RAIM Test Statistic during period of High PRN01 Error on 10th July 2023 at SLGO

It is noted that this error is linked with an unscheduled outage (NANU 2023036) that is raised on 10th July. In this in states that PRN01 is unusable after 10:20. The reason why PRN01 is removed from the solution around this time is that observations for it no longer appear in the logged data.

It is possible that such an error could be a local fault (e.g. multipath or non-line-of-sight error), or it could be that there is a satellite issue but the post-processing uses an incorrect navigation message. To check these we look at results from other sites and real-time results from COTS receivers.

Firstly, at GMV in Nottingham, there are multiple receivers running continuously and generating real time solutions. These will be using the signals and navigation messages as broadcast at that time by the satellites. If these also see an error that suggests that the fault is real and not simply an effect of incorrect usage of navigation messages in post-processing. The plots below show the real-time position errors from the receiver solutions for a selection of receivers. Note that in all cases, the x-axis corresponds to time since 09:00 and so 0 is 09:00, 60 is 10:00, etc.



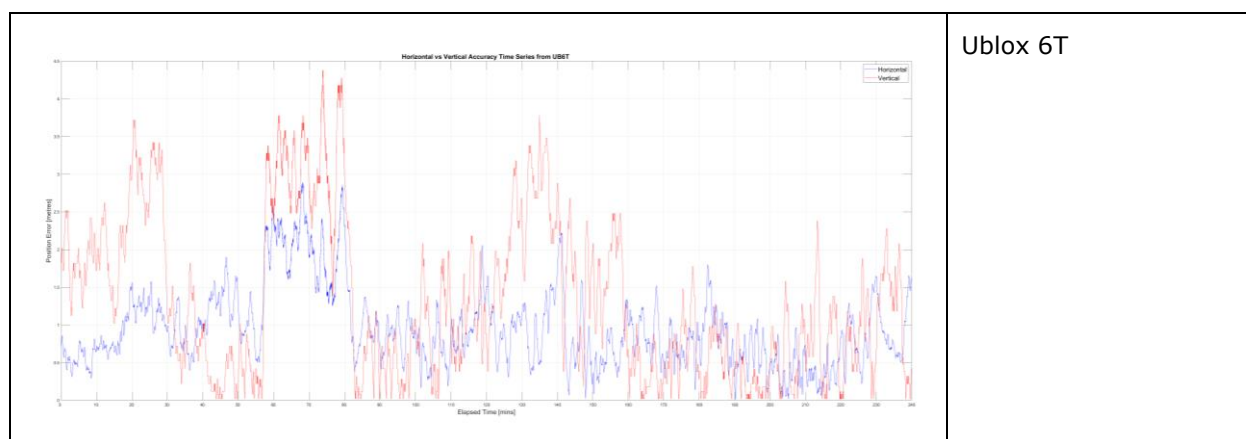


Figure 5-4: Position Errors for COTS Receiver Real-Time Solutions on 10th July 2023 at Nottingham

From these results we can see that some of the real-time solutions also show increased position errors at the same time as the post-processed solution used for the performance assessment. This points to the satellite range error (and resultant position error) being a real error that would have affected receivers in real-time. It is also noted that not all the receivers show increased errors – perhaps because of internal data checks that identify and remove faulty measurements before using them.

The fact that the error is seen in real-time receiver solutions and in post-processing and is seen at different locations with high range residuals for the same satellite demonstrates that this was a real SIS integrity fault affecting PRN01.

In terms of system alerting, NANU 202306 was raised at 10:20 – i.e. approx. 23 minutes after the start of the problems – and an updated navigation message (indicating unhealthy status for the satellite) was broadcast at around 11:06 (within 1hr 9 mins of the start of the issue).

22ND AUGUST 2023

On 22nd August 2023, the range residual errors measured on satellite SVN55 / PRN15 exceed the threshold ($4.42 \times \text{URA}$) for 11 seconds at just after 17:50, resulting in increased position errors. For the entire period, RAIM FD flags there is a fault.

The range error is around 25m on PRN15, which appears in the observation file suddenly at 35 degrees elevation – before this time there are no measurements for that satellite for many hours.

This increase in range error for PRN15 affects the position solution (both horizontal and vertical) so that the error is larger than usual during this period, although not excessive. This is shown in the next figure.

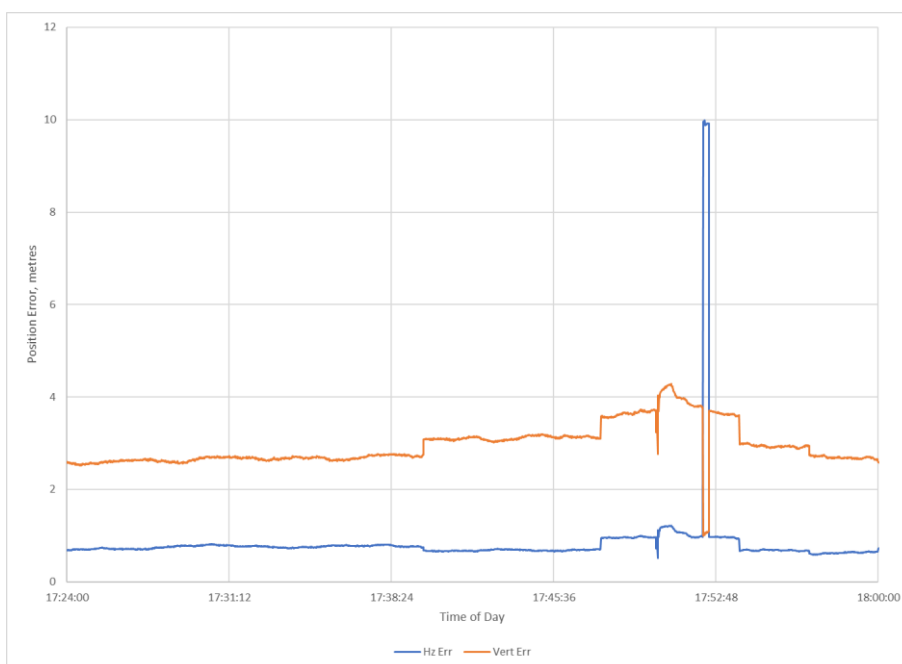


Figure 5-5: Position Error during period of High PRN15 Error on 22nd August 2023 at SLGO

Next, we can see that it is possible to detect this error though RAIM FD. In this figure the weighted least squares RAIM test statistic (which is a measure of the consistency of the residuals) is shown, along with the detection threshold.

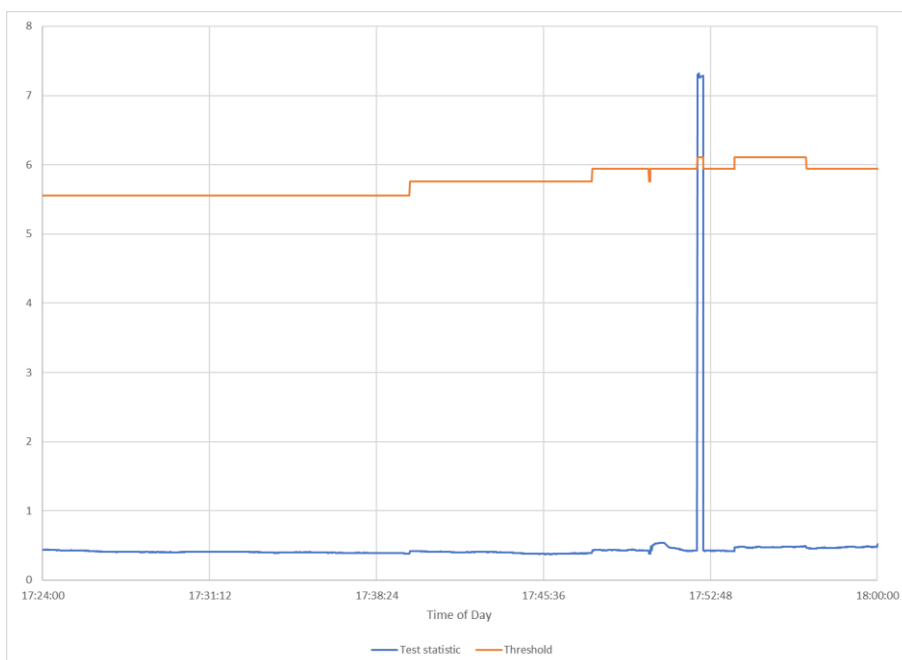


Figure 5-6: RAIM Test Statistic during period of High PRN15 Error on 22nd August 2023 at SLGO

It is noted that this error is linked with an unscheduled outage (NANU 2023047) that is raised on 22nd August. In this in states that PRN15 is unusable after 16:11, which is before the time when we see this fault. However, what seems to happen is the following:

- Until 16:11, the satellite is 'normal' , although it is not in view at SLGO at this time (it is below the horizon),
- Nevertheless, the combined navigation message product used in post-processing (created based on navigation messages received at a global network of GNSS receivers) includes the last valid nav message broadcast from 16:00,
- At 16:11, the GPS system decides PRN15 is unhealthy and so stops transmitting and raises a NANU,
- At 17:52:19, PRN15 starts transmitting a signal again that is now logged at SLGO,
- The satellite is used in the solution with the last valid nav message that was received,
- At 17:52:30 a new nav message is received with 'unhealthy' status for PRN15 and so it is then removed from the solution.

If we look at the real-time solutions from the COTS receivers in Nottingham then they do not show this large position error. This is because they do not use PRN15 in the solution at this time – even though they are tracking it. The reason for that is that in real-time they do not have access to the previous healthy navigation message because it was not in view at Nottingham when the last healthy navigation message was broadcast. In fact, there is no location in Ireland where PRN15 was in view when the fault started and so real-time receivers should not have used it in the solution when it returned. The post-processing indicates there was a potential problem though, and in fact any receiver that was at UK latitudes out in the Atlantic (further than 16 degrees West) would have tracked PRN15 when it was still healthy and when it returned unhealthy and may have suffered the same short error as was seen in post-processing.



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